In 1666, while London was racked by the great fire, a much more significant event took place 100 miles North, as Isaac Newton propounded

Newton's Law of Gravitation: Two particles of masses m_0 and m_1 separated by a distance r are attracted towards each other by a force of magnitude

$$F = \frac{Gm_0m_1}{r^2}$$
 where G is the gravitational constant.

Of the various physical constants which define how the universe works, G is perhaps the hardest to measure accurately, but $G \simeq 6.67 \times 10^{-11}$ in standard units. (What are its physical dimensions?) The acceleration of particle m_1 due to the presence of m_0 is thus the same whatever its mass. There is actually a subtle question here, as one can distinguish two kinds of mass, "inertial mass" (as in $\mathbf{F} = m\mathbf{a}$) measuring how hard it is to cause a particle to start moving, and gravitational mass (how hard a particle pulls at another.) It seems to be a physical fact that these are the same.

The gravitational potential, V, due to a mass M is thus

$$V(r) = -\frac{GMm}{r}$$
 so that $V_{\rm eff}(r) = \frac{mh^2}{2r^2} - \frac{GMm}{r}$

where $h = r^2 \dot{\theta}$ is the angular momentum per unit mass. The orbit of the particle m is given by the energy equation

$$\frac{1}{2}m\dot{r}^2 + V_{\text{eff}} = E$$
 a constant.

Drawing V_{eff} against r, we see that negative values of E correspond to oscillatory behaviour, whereas if E > 0, the mass m escapes to infinity.

Planetary orbits: Returning to the orbit equation (3.7), we see that a planet of mass m orbiting the sun of mass M feels a force

$$F(r) = -\frac{GMm}{r^2}$$
 so that $\frac{d^2u}{d\theta^2} + u = \frac{GM}{h^2} \equiv \frac{1}{l}$ (3.9)

where l is constant. The particular integral for u is u = 1/l, so the general solution is

$$u = \frac{1}{I} [1 + e \cos(\theta + \phi)]$$
 for constant ϕ and e .

Without loss of generality, $e \ge 0$, while for any particular orbit we can choose the definition of θ in such a way that $\phi = 0$. This leaves us with the standard orbit equation

$$\frac{l}{r} = 1 + e\cos\theta. \tag{3.10}$$

This describes a curve in the (r, θ) -plane. What does it look like? First, note that r is a 2π -periodic function of θ , so that a bounded orbit must be closed. When e = 0, r = l and

we have a circle. If 0 < e < 1, l/r oscillates between (1 - e) and (1 + e) so that the orbit is bounded. If $e \ge 1$, however, $r \to \infty$ as $\cos \theta \to -1/e$, and the planet escapes to infinity. We can rewrite (3.10) in Cartesian coordinates,

$$l = r + er \cos \theta = \sqrt{x^2 + y^2} + ex$$
 so that $l - ex = \sqrt{x^2 + y^2}$.

Squaring, and rearranging, we have

$$x^{2}(1-e^{2}) + 2elx + y^{2} = l^{2}$$
 or $\left(x + \frac{el}{1-e^{2}}\right)^{2} + \frac{y^{2}}{1-e^{2}} = \frac{l^{2}}{(1-e^{2})^{2}}$.

We recognise this curve as a **conic**. When e = 0 it is a circle, when 0 < e < 1 it is an ellipse, when e = 1 it is a parabola and when e > 1 it is a hyperbola. The parameter e is called the **eccentricity** of the orbit. All planets, asteroids, comets etc. have elliptical orbits, if we ignore the interactions between them. Comparing with the standard equation for an ellipse, $X^2/a^2 + Y^2/b^2 = 1$, we see that

$$a = \frac{l}{1 - e^2}, \quad b = \frac{l}{\sqrt{1 - e^2}}, \quad e = \left(1 - \frac{b^2}{a^2}\right)^{1/2}, \quad l = \frac{b^2}{a}.$$
 (3.11)

The centre of the ellipse is at x = -ae, so that the sun is at a **focus** of the ellipse.

Kepler's laws: Newton's theory was to a large extent deduced from observations by Tycho de Brahe interpreted by Johann Kepler, whose two laws from 1609 and a third from 1618, described how the planets move:

- (1) The orbits are ellipses with the sun at one of the foci.
- (2) The radius vector from the sun sweeps out equal areas in equal times.
- (3) The square of the orbital period of each planet is proportional to the cube of the semi-major axis (a, above), and the constant of proportionality is the same for each planet.

We have verified that Newton's inverse-square-law for gravitation agrees with the first of Kepler's laws. The second holds for any **central force**, as it is essentially angular momentum conservation. The area $A(\theta)$ of a curve $r = r(\theta)$ can be written as $A = \frac{1}{2} \int_0^{\theta} r^2 d\theta$ so that $dA/dt = \frac{1}{2}r^2\dot{\theta} = \frac{1}{2}h$. Kepler's third law then follows from the first two. The orbital period (or 'year'), τ , is the time taken for a complete revolution, during which time an entire ellipse is swept out by the radius. Thus $\tau = 2A_e/h$, where $A_e = \pi ab$ is the area of an ellipse. Using (3.11) and (3.9), we have

$$\tau^2 = \frac{4\pi^2 a^2 b^2}{h^2} = \frac{4\pi^2 a^2 b^2}{GMl} = \frac{4\pi^2}{GM} a^3.$$

Some astronomical data: The sun's mass is 1.99×10^{30} kg, while the earth's is 5.98×10^{24} kg and the moon's 7.34×10^{22} kg. The eccentricity of the earth's orbit is small, e = 0.017. Most planets have approximately circular orbits, but for Pluto $e \simeq 0.25$ and Mercury $e \simeq 0.2$. As a result, in 1999, Pluto was nearer the sun than Neptune! Comets have highly elliptical orbits, and so take a long time to return; Halley's comet, for example has e = 0.967. The earth is 1.49×10^{11} m from the sun, and 3.84×10^{8} m from the moon. Its radius is 6.37×10^{6} m.